

AA216/CME345: MODEL REDUCTION

Local Parametric Approaches

Charbel Farhat
Stanford University
cfarhat@stanford.edu

Outline

- 1** Concept of a Database of Local PROMs
- 2** Interpolation of ROBs on Quotient Manifolds (Amsallem and Farhat, 2008)
- 3** Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

- Note: The material covered in this chapter is based on the following published papers:
 - D. Amsallem, C. Farhat. Interpolation method for adapting reduced-order models and application to aeroelasticity. *AIAA Journal* 2008; 46(7):1803–1813.
 - D. Amsallem, J. Cortial, C. Farhat. Towards real-time CFD-based aeroelastic computations using a database of reduced-order models. *AIAA Journal* 2010; 48(9):2029-2037.
 - D. Amsallem, C. Farhat. An online method for interpolating linear parametric reduced-order models. *SIAM Journal on Scientific Computing* 2011; 33(5): 2169-2198.
 - D. Amsallem, Interpolation on manifolds of CFD-based fluid and finite element-based structural reduced-order models for on-line aeroelastic predictions. Ph.D. Thesis, Stanford University, 2010.

$$\begin{aligned}\frac{d\mathbf{w}}{dt}(t; \boldsymbol{\mu}) &= \mathbf{A}(\boldsymbol{\mu})\mathbf{w}(t; \boldsymbol{\mu}) + \mathbf{B}(\boldsymbol{\mu})\mathbf{u}(t) \\ \mathbf{y}(t; \boldsymbol{\mu}) &= \mathbf{C}(\boldsymbol{\mu})\mathbf{w}(t; \boldsymbol{\mu}) + \mathbf{D}(\boldsymbol{\mu})\mathbf{u}(t) \\ \mathbf{w}(0; \boldsymbol{\mu}) &= \mathbf{w}_0(\boldsymbol{\mu})\end{aligned}$$

- $\mathbf{w} \in \mathbb{R}^N$: Vector of state variables
- $\mathbf{u} \in \mathbb{R}^p$: Vector of input variables – typically $p \ll N$
- $\mathbf{y} \in \mathbb{R}^q$: Vector of output variables – typically $q \ll N$
- $\boldsymbol{\mu} \in \mathcal{D} \subset \mathbb{R}^m$: Vector of parameters – typically $m \ll N$

- Goal: Construct a parametric Projection-based Reduced-Order Model (PROM)

$$\begin{aligned}\frac{d\mathbf{q}}{dt}(t; \boldsymbol{\mu}) &= \mathbf{A}_r(\boldsymbol{\mu})\mathbf{q}(t; \boldsymbol{\mu}) + \mathbf{B}_r(\boldsymbol{\mu})\mathbf{u}(t) \\ \mathbf{y}(t; \boldsymbol{\mu}) &= \mathbf{C}_r(\boldsymbol{\mu})\mathbf{q}(t; \boldsymbol{\mu}) + \mathbf{D}_r(\boldsymbol{\mu})\mathbf{u}(t)\end{aligned}$$

- based on **local** Reduced-Order Bases (ROBs) $(\mathbf{V}(\boldsymbol{\mu}^{(l)}), \mathbf{W}(\boldsymbol{\mu}^{(l)}))$ and the approximation

$$\mathbf{w}(t; \boldsymbol{\mu}) \approx \mathbf{V}(\boldsymbol{\mu})\mathbf{q}(t; \boldsymbol{\mu})$$

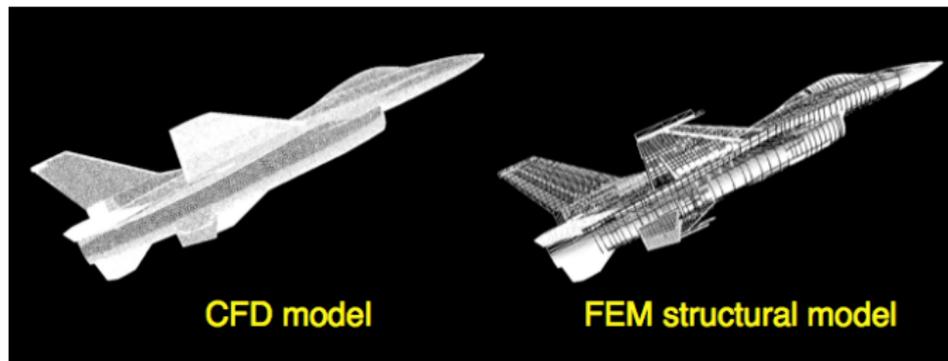
- $\boldsymbol{\mu}^{(l)} \in \mathcal{D}$; $\mathbf{q} \in \mathbb{R}^k$
- all ROBs have the same dimension $k \ll N$
- PROM resulting from Petrov-Galerkin projection

$$\mathbf{A}_r(\boldsymbol{\mu}) = (\mathbf{W}(\boldsymbol{\mu})^T \mathbf{V}(\boldsymbol{\mu}))^{-1} \mathbf{W}(\boldsymbol{\mu})^T \mathbf{A}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}) \in \mathbb{R}^{k \times k}$$

$$\mathbf{B}_r(\boldsymbol{\mu}) = (\mathbf{W}(\boldsymbol{\mu})^T \mathbf{V}(\boldsymbol{\mu}))^{-1} \mathbf{W}(\boldsymbol{\mu})^T \mathbf{B}(\boldsymbol{\mu}) \in \mathbb{R}^{k \times p}$$

$$\mathbf{C}_r(\boldsymbol{\mu}) = \mathbf{C}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}) \in \mathbb{R}^{q \times k}; \quad \mathbf{D}_r(\boldsymbol{\mu}) = \mathbf{D}(\boldsymbol{\mu}) \in \mathbb{R}^{q \times p}$$

- Hundreds of flight conditions $\mu = (M_\infty, \alpha)$ for flutter clearance
- Multiple aircraft configurations



- $N_{\text{fluid}} \approx 2 \times 10^6$, $N_{\text{structure}} \approx 1.6 \times 10^5$

└ Concept of a Database of Local PROMs

└ Lack of Robustness of Local ROBs for Parameter Changes

- Consider the following procedure
 - 1 construct local ROBs $\left(\mathbf{V} \left(\mu^{(1)} \right), \mathbf{W} \left(\mu^{(1)} \right) \right)$ at the parametric configuration/flight condition $\mu^{(1)}$

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- use these two bases to reduce the HDM at $\boldsymbol{\mu}^{(2)}$
- avoid reconstructing a ROB every time the configuration/flight condition is varied
- build the resulting PROM

$$\begin{aligned}\frac{d\mathbf{q}}{dt}(t; \boldsymbol{\mu}^{(2)}) &= \mathbf{A}_r(\boldsymbol{\mu}^{(2)}) \mathbf{q}(t; \boldsymbol{\mu}^{(2)}) + \mathbf{B}_r(\boldsymbol{\mu}^{(2)}) \mathbf{u}(t) \\ \mathbf{y}(t; \boldsymbol{\mu}^{(2)}) &= \mathbf{C}_r(\boldsymbol{\mu}^{(2)}) \mathbf{q}(t; \boldsymbol{\mu}^{(2)}) + \mathbf{D}_r(\boldsymbol{\mu}^{(2)}) \mathbf{u}(t) \\ \mathbf{w}(t, \boldsymbol{\mu}^{(2)}) &\approx \mathbf{V}(\boldsymbol{\mu}^{(1)}) \mathbf{q}(t; \boldsymbol{\mu}^{(2)})\end{aligned}$$

where

$$\mathbf{A}_r(\boldsymbol{\mu}^{(2)}) = \left(\mathbf{W}(\boldsymbol{\mu}^{(1)})^T \mathbf{V}(\boldsymbol{\mu}^{(1)}) \right)^{-1} \mathbf{W}(\boldsymbol{\mu}^{(1)})^T \mathbf{A}(\boldsymbol{\mu}^{(2)}) \mathbf{V}(\boldsymbol{\mu}^{(1)}) \in \mathbb{R}^{k \times k}$$

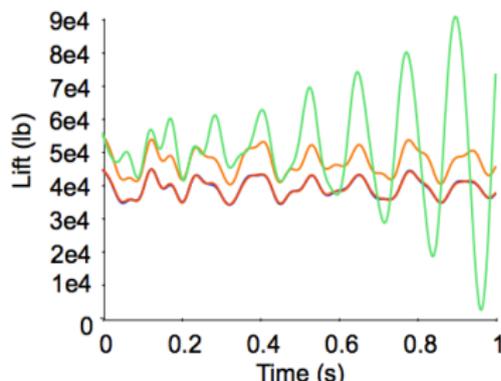
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- Queried flight conditions

- $\mu^{(1)} = (M_{\infty,1}, \alpha^{(1)}) = (0.71, \alpha_{\text{trimmed}}(0.71))$

- $\mu^{(2)} = (M_{\infty,2}, \alpha^{(2)}) = (0.8, \alpha_{\text{trimmed}}(0.8))$



- $HDM(\mu^{(1)})$
- $PROM(\mu^{(1)})$
- $HDM(\mu^{(2)})$
- $PROM(\mu^{(2)})$

⇒ the ROBs lack robustness with respect to parameter changes

- The lack of robustness of the ROBs with respect to parameter changes implies that the ROBs should be **reconstructed** every time the parameters are varied
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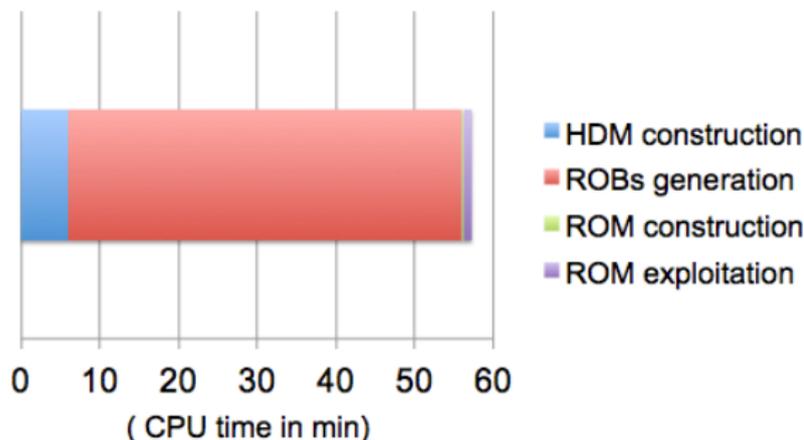
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- Question: Is this procedure computationally **efficient**?

- Construction and exploitation in $t \in [0, 1]$ s of a linearized aeroelastic F-16 PROM



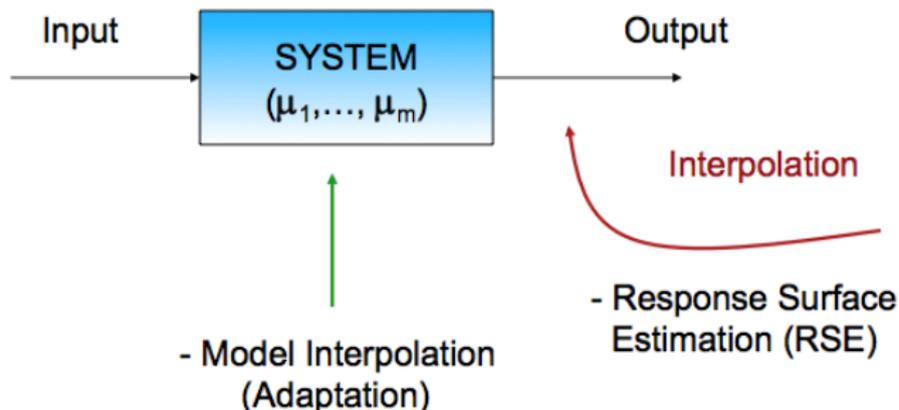
- The direct generation of the ROB accounts for 89% of the total CPU time
- The overall procedure takes 56 minutes, which renders this approach **non-amenable to real-time parametric applications**

└ Concept of a Database of Local PROMs

└ One Strategy

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- What entities should be interpolated and how?



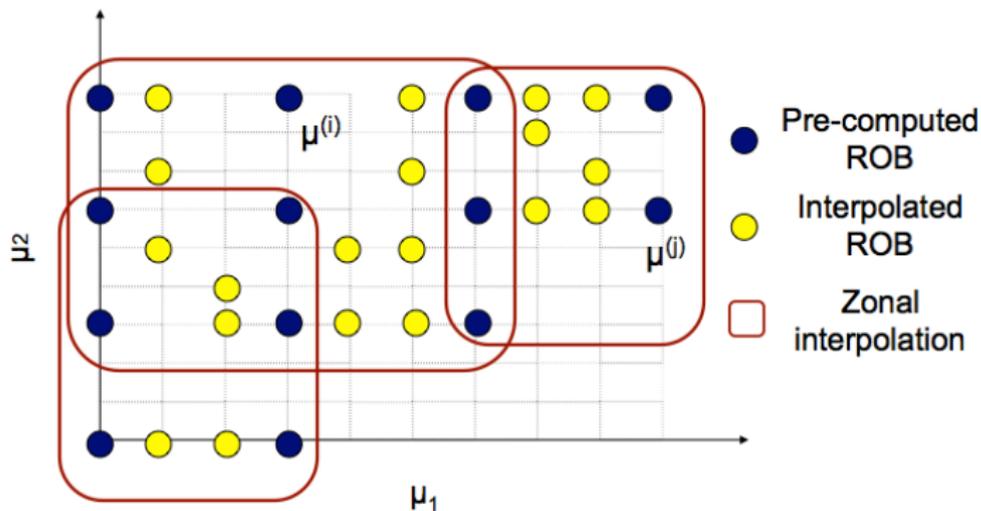
⇒ interpolate the ROBs given the above CPU time analysis

■ Idea

- pre-compute ROBs at a number of sampled parameter points

$$\left\{ \mu^{(l)} \in \mathcal{D} \right\}_{l=1}^{N_s}$$

- interpolate these ROBs to obtain a ROB at a queried but unsampled parameter configuration $\mu^* \notin \left\{ \mu^{(l)} \right\}_{l=1}^{N_s}$



- For simplicity, assume an orthogonal Galerkin projection

$$\mathbf{V}(\boldsymbol{\mu}) = \mathbf{W}(\boldsymbol{\mu}) \quad \text{and} \quad \mathbf{V}(\boldsymbol{\mu})^T \mathbf{V}(\boldsymbol{\mu}) = \mathbf{I}_k$$

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- 4 construct the PROM operators $(\mathbf{A}_r(\boldsymbol{\mu}^*), \mathbf{B}_r(\boldsymbol{\mu}^*), \mathbf{C}_r(\boldsymbol{\mu}^*), \mathbf{D}_r(\boldsymbol{\mu}^*))$ by Galerkin projection

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- Question: How does one interpolate pre-computed ROBs?

└ Interpolation of ROBs on Quotient Manifolds (Amsallem and Farhat, 2008)

└ Direct Interpolation of the ROBs

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 - pre-computed ROBs $\left\{ \mathbf{V}(\boldsymbol{\mu}^{(l)}) \right\}_{l=1}^{N_s}$
 - multi-variate interpolation in \mathbb{R}^m (operator \mathcal{I})

$$a(\boldsymbol{\mu}) = \mathcal{I} \left(\boldsymbol{\mu}; \left\{ a(\boldsymbol{\mu}^{(l)}) \right\}_{l=1}^{N_s}, \boldsymbol{\mu}^{(l)} \right)$$

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- Algorithm

- 1: **for** $i = 1 : N$ **do**
- 2: **for** $j = 1 : k$ **do**
- 3: compute $v_{ij}(\boldsymbol{\mu}^*) = \mathcal{I}\left(\boldsymbol{\mu}^*; \left\{v_{ij}(\boldsymbol{\mu}^{(l)}), \boldsymbol{\mu}^{(l)}\right\}_{l=1}^{N_s}\right)$
- 4: **end for**
- 5: **end for**
- 6: form $\mathbf{V}(\boldsymbol{\mu}^*) = [v_{ij}(\boldsymbol{\mu}^*)]$

■ Example

- $N = 3, k = 2, m = 1$
- for $\mu^{(1)} = 0$: $\mathbf{V}(\mu^{(1)}) = \mathbf{V}(0) = (\mathbf{v}_1 \ \mathbf{v}_2)^T$
- for $\mu^{(2)} = 1$: $\mathbf{V}(\mu^{(2)}) = \mathbf{V}(1) = (-\mathbf{v}_1 \ \mathbf{v}_2)^T$
- target parameter $\mu = 0.5$
- linear interpolation

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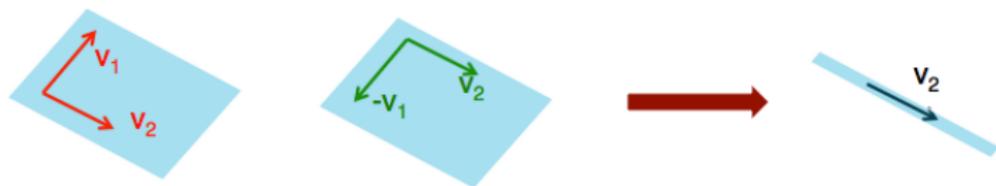
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■ Interpolatory result

$$\mathbf{V}(0.5) = 0.5(\mathbf{V}(0) + \mathbf{V}(1)) = (0.5(\mathbf{v}_1 - \mathbf{v}_1) \ 0.5(\mathbf{v}_2 + \mathbf{v}_2))^T = (0 \ \mathbf{v}_2)^T$$



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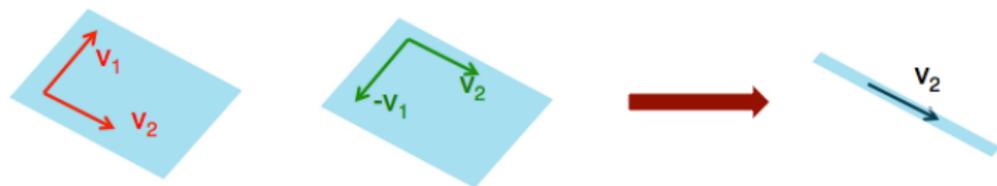
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■ What went wrong?

- a **relevant constraint** was neither identified nor preserved

- the **wrong entity** was interpolated

- Reduced-order equation

$$\frac{d\mathbf{q}}{dt}(t; \boldsymbol{\mu}) = \mathbf{V}(\boldsymbol{\mu})^T \mathbf{A}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}) \mathbf{q}(t; \boldsymbol{\mu}) + \mathbf{V}(\boldsymbol{\mu})^T \mathbf{B}(\boldsymbol{\mu}) \mathbf{u}(t)$$

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- Equivalent high-dimensional equation for $\tilde{\mathbf{w}}(t; \boldsymbol{\mu}) = \mathbf{V}(\boldsymbol{\mu}) \mathbf{q}(t; \boldsymbol{\mu})$

$$\frac{d\tilde{\mathbf{w}}}{dt}(t; \boldsymbol{\mu}) = \boldsymbol{\Pi}_{\mathbf{V}(\boldsymbol{\mu}), \mathbf{V}(\boldsymbol{\mu})} \mathbf{A}(\boldsymbol{\mu}) \tilde{\mathbf{w}}(t; \boldsymbol{\mu}) + \boldsymbol{\Pi}_{\mathbf{V}(\boldsymbol{\mu}), \mathbf{V}(\boldsymbol{\mu})} \mathbf{B}(\boldsymbol{\mu}) \mathbf{u}(t)$$

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- The PROM solution is independent of the choice of ROB associated with the projection subspace

⇒ the **correct entity to interpolate** is $\mathcal{S}(\boldsymbol{\mu}) = \text{range}(\mathbf{V}(\boldsymbol{\mu}))$

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- Manifolds of interest
 - $\mathcal{G}(k, N)$ (Grassmann manifold): Set of subspaces in \mathbb{R}^N of dimension k
 - $ST(k, N)$ (orthogonal Stiefel manifold): Set of orthogonal ROB matrices in $\mathbb{R}^{N \times k}$
 - $\text{GL}(k)$ (general linear group): Set of non-singular square matrices of size k
 - $\mathcal{O}(k)$: Set of orthogonal square matrices of size k

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- Properties
 - $\mathcal{O}(k) \subset \text{GL}(k)$
 - $ST(N, N) = \mathcal{O}(N)$

- Case of projection-based model order reduction with orthogonal ROBs
 - $\mathbf{V}(\boldsymbol{\mu}) \in \mathcal{ST}(k, N)$
 - $\text{range}(\mathbf{V}(\boldsymbol{\mu})) \in \mathcal{G}(k, N)$

└ The Grassmann Manifold

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 - $\mathbf{V}(\boldsymbol{\mu}) \in \mathcal{ST}(k, N)$
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 - this class of equivalence is defined by the range operation

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└ The Grassmann Manifold

- Case of projection-based model order reduction with orthogonal ROBs

- $\mathbf{V}(\boldsymbol{\mu}) \in \mathcal{ST}(k, N)$
- $\text{range}(\mathbf{V}(\boldsymbol{\mu})) \in \mathcal{G}(k, N)$

- Equivalence class

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- Hence, one should interpolate subspaces, but has access in practice to (orthogonal) ROBs
- Solution: Perform interpolation on the Grassmann manifold using entities belonging to the (orthogonal) Stiefel manifold

└ Matrix Manifolds

■ Embedded matrix manifolds

- the sphere

$$\mathbb{S}(N) = \left\{ \mathbf{w} \in \mathbb{R}^N \text{ s.t. } \|\mathbf{w}\|_2 = 1 \right\}$$

- the manifold of orthogonal matrices

$$\mathcal{O}(N) = \left\{ \mathbf{M} \in \mathbb{R}^{N \times N} \text{ s.t. } \mathbf{M}^T \mathbf{M} = \mathbf{I}_N \right\}$$

- the general linear group

$$\text{GL}(N) = \left\{ \mathbf{M} \in \mathbb{R}^{N \times N} \text{ s.t. } \det(\mathbf{M}) \neq 0 \right\}$$

- the manifold of symmetric positive definite matrices

$$\text{SPD}(N) = \left\{ \mathbf{M} \in \mathbb{R}^{N \times N} \text{ s.t. } \mathbf{M} = \mathbf{M}^T \& \mathbf{w}^T \mathbf{M} \mathbf{w} > 0 \forall \mathbf{w} \neq \mathbf{0} \right\}$$

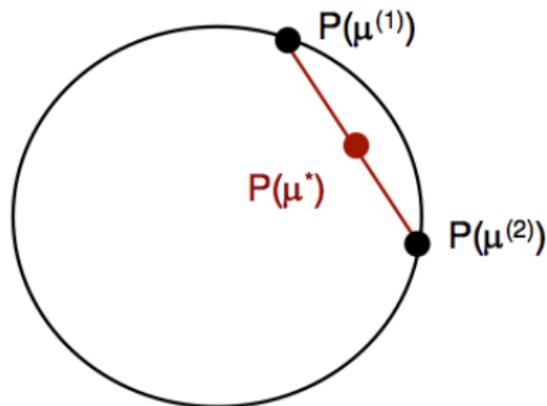
- the orthogonal Stiefel manifold

$$\text{ST}(k, N) = \left\{ \mathbf{M} \in \mathbb{R}^{N \times k} \text{ s.t. } \mathbf{M}^T \mathbf{M} = \mathbf{I}_k \right\}$$

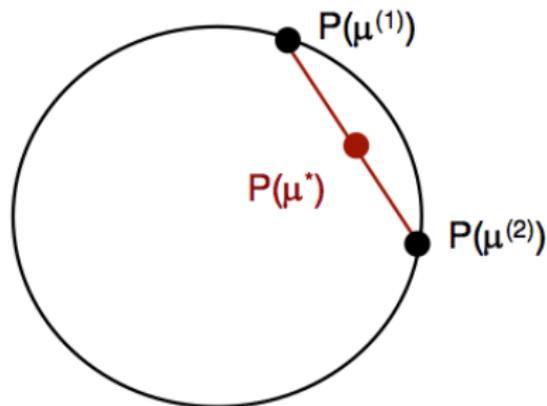
■ Quotient matrix manifold

- the Grassmann manifold

- First example: The circle (sphere $\mathbb{S}(N)$ for $N = 2$)

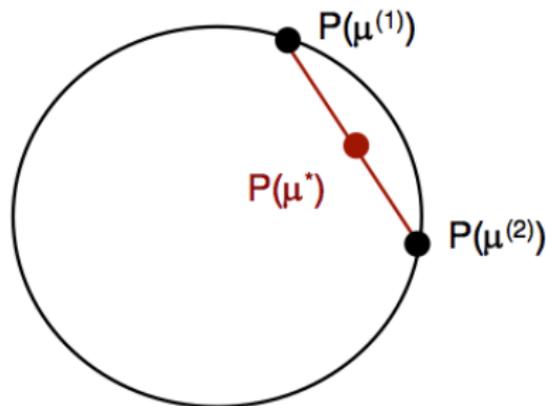


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- Standard interpolation fails to preserve a nonlinear manifold (essentially because standard interpolation applies only in vector spaces)
- Idea: perform interpolation in a linear space \Rightarrow on a **tangent space to the manifold**

- Input

- pre-computed matrices $\left\{ \mathbf{A}(\boldsymbol{\mu}^{(l)}) \in \mathbb{R}^{N \times M} \right\}_{l=1}^{N_s}$
- map $m_{\mathbf{A}}$ from the manifold \mathcal{M} to the tangent space of \mathcal{M} at \mathbf{A}
- multi-variate interpolation in \mathbb{R}^m

$$\left(\text{operator } a(\boldsymbol{\mu}) = \mathcal{I} \left(\boldsymbol{\mu}; \left\{ a(\boldsymbol{\mu}^{(l)}), \boldsymbol{\mu}^{(l)} \right\}_{l=1}^{N_s} \right) \right)$$
- inverse map $m_{\mathbf{A}}^{-1}$ from the tangent space to \mathcal{M} at \mathbf{A} to the manifold \mathcal{M}

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- Requirement: The interpolation operator \mathcal{I} must preserve the tangent space \Rightarrow for example,

$$a(\boldsymbol{\mu}^*) = \mathcal{I} \left(\boldsymbol{\mu}^*; \left\{ a \left(\boldsymbol{\mu}^{(l)} \right), \boldsymbol{\mu}^{(l)} \right\}_{l=1}^{N_s} \right) = \sum_{l=1}^{N_s} \theta_l(\boldsymbol{\mu}^*) a \left(\boldsymbol{\mu}^{(l)} \right)$$

■ Algorithm

- 1: **for** $l = 1 : N_s$ **do**
- 2: compute $\Gamma(\mu^{(l)}) = m_{\mathbf{A}}(\mathbf{A}(\mu^{(l)}))$
- 3: **end for**
- 4: **for** $i = 1 : N$ **do**
- 5: **for** $j = 1 : M$ **do**
- 6: compute $\Gamma_{ij}(\mu^*) = \mathcal{I}(\mu^*; \{\Gamma_{ij}(\mu^{(l)}), \mu^{(l)}\}_{l=1}^{N_s})$
- 7: **end for**
- 8: **end for**
- 9: form $\Gamma(\mu^*) = [\Gamma_{ij}(\mu^*)]$ and compute $\mathbf{A}(\mu^*) = m_{\mathbf{A}}^{-1}(\Gamma(\mu^*))$

- How does one find $m_{\mathbf{A}}$ and its inverse $m_{\mathbf{A}}^{-1}$?

└ Interpolation of ROBs on Quotient Manifolds (Amsallem and Farhat, 2008)

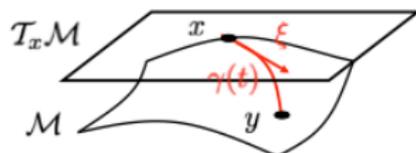
└ Differential Geometry

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 - is a generalization of a “straight line” to “curved spaces” (manifolds)
 - is uniquely defined given a point x on the manifold and a tangent vector χ at this point

$$\gamma(t; x, \xi) : [0, 1] \rightarrow \mathcal{M}$$

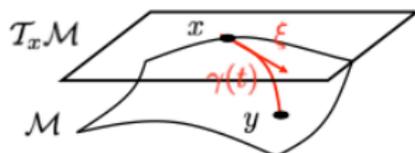
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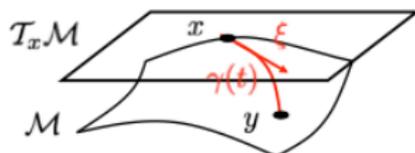
- Exponential map

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- Exponential map

$$\text{Exp}_x : T_x \mathcal{M} \rightarrow \mathcal{M} \quad \xi \mapsto \gamma(1; x, \xi)$$

- Logarithmic map (defined in a neighborhood \mathcal{U}_x of x)

$$\text{Log}_x : \mathcal{U}_x \subset \mathcal{M} \rightarrow T_x \mathcal{M} \quad y \mapsto \text{Exp}_x^{-1}(y) = \text{Log}_x(y) = \dot{\gamma}(0, x, \xi) = \xi$$

■ Logarithmic map

- 1 compute a thin SVD

$$(\mathbf{I} - \mathbf{V}_0 \mathbf{V}_0^T) \mathbf{V}_i (\mathbf{V}_0^T \mathbf{V}_i)^{-1} = \mathbf{U} \mathbf{\Sigma} \mathbf{Z}^T$$

- 2 compute

$$\mathbf{\Gamma} = \mathbf{U} \tan^{-1}(\mathbf{\Sigma}) \mathbf{Z}^T \in \mathbb{R}^{N \times k}$$

- 3 $\mathbf{\Gamma} \leftrightarrow \text{Log}_{S_0}(S_i) \in \mathcal{T}_{S_0} \mathcal{G}(k, N)$

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- Exponential map of $\tilde{\chi} \in \mathcal{T}_{S_0} \mathcal{G}(k, N) \leftrightarrow \tilde{\mathbf{\Gamma}}$

- 1 compute a thin SVD

$$\tilde{\mathbf{\Gamma}} = \mathbf{U} \mathbf{\Sigma} \mathbf{Z}^T$$

- 2 compute

$$\mathbf{V} = (\mathbf{V}_0 \mathbf{Z} \cos \mathbf{\Sigma} + \mathbf{U} \sin \mathbf{\Sigma}) \in \mathcal{ST}(k, N)$$

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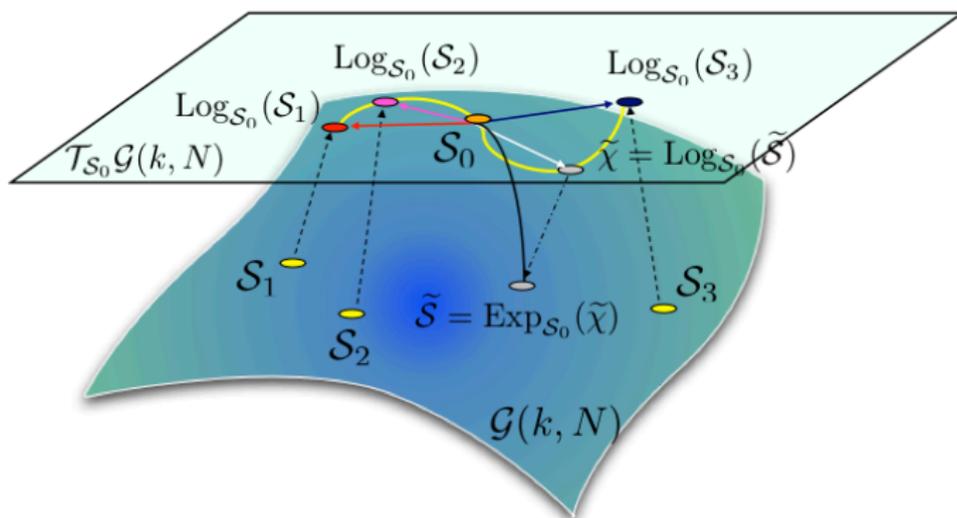
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- Note: The trigonometric operators apply only to the diagonal entries of the relevant matrices

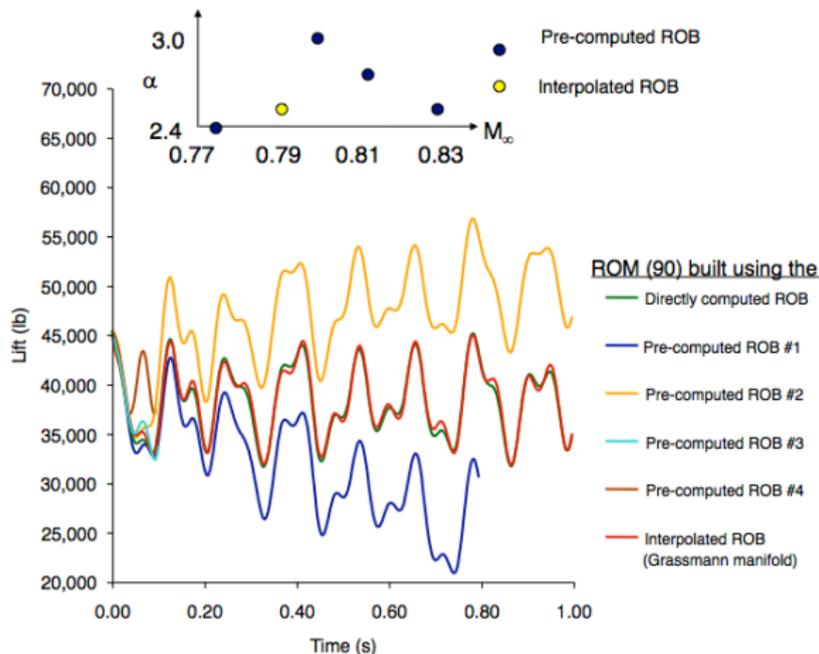
- Interpolation of ROBs on Quotient Manifolds (Amsallem and Farhat, 2008)

- Interpolation on a Tangent Space to the Grassmann Manifold

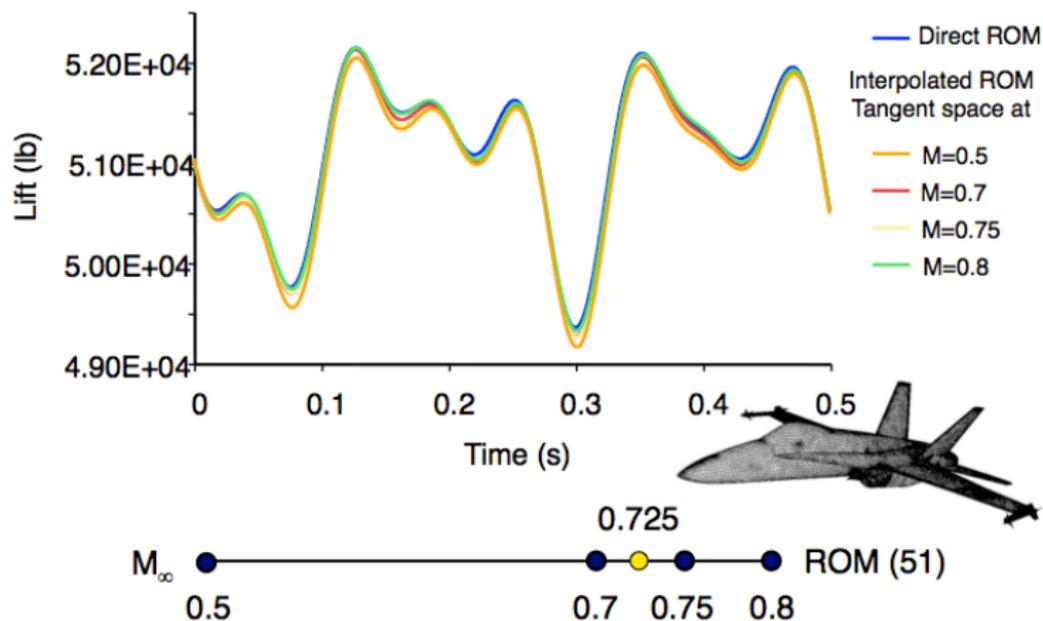
- Interpolation on the tangent space to $\mathcal{G}(k, N)$



- Prediction of the linearized aeroelastic behavior of an F-16 configuration



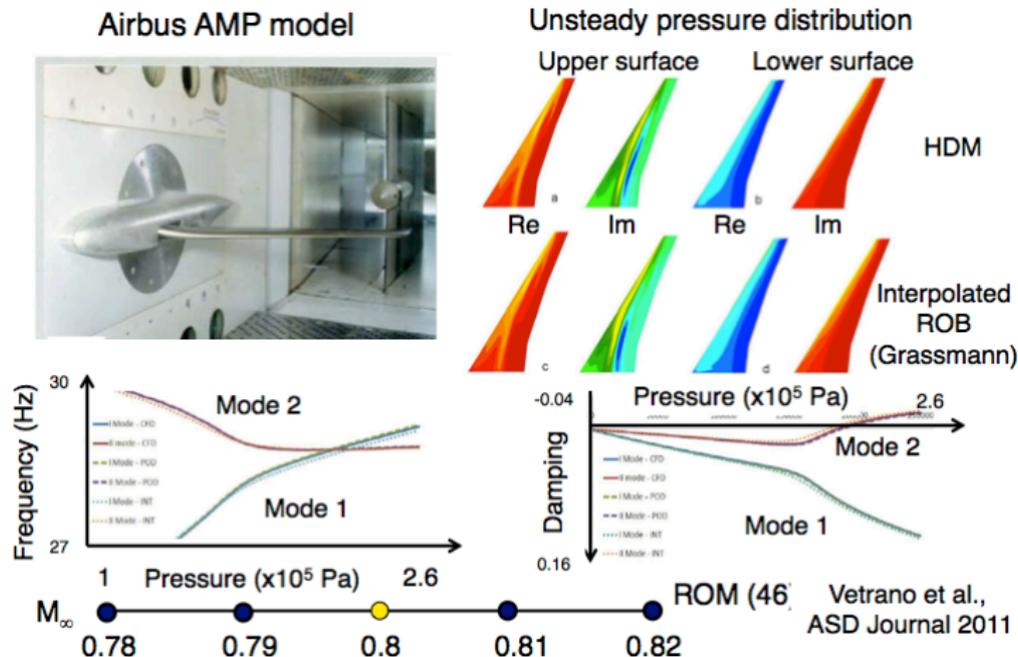
- Prediction of the linearized aeroelastic behavior of an F-18 configuration: Effect of the choice of the tangent plane



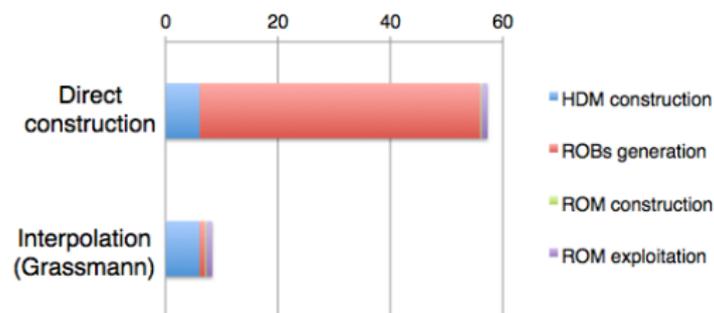
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- Application to Linearized Aeroelasticity

- Prediction of the linearized aeroelastic behavior of the wing of a commercial aircraft (Airbus)

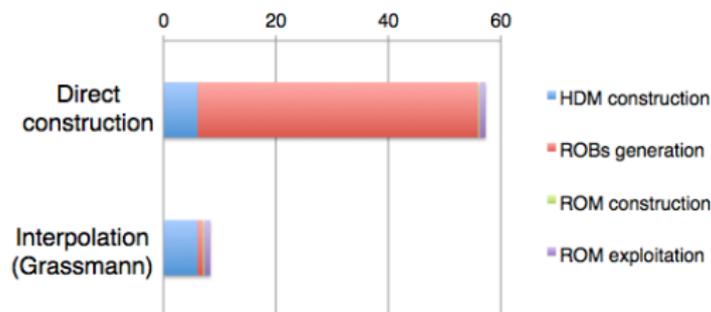


- Construction and exploitation in $t \in [0, 1]$ s of a linearized aeroelastic PROM



- Overall CPU time is decreased from 55 minutes to 8 minutes

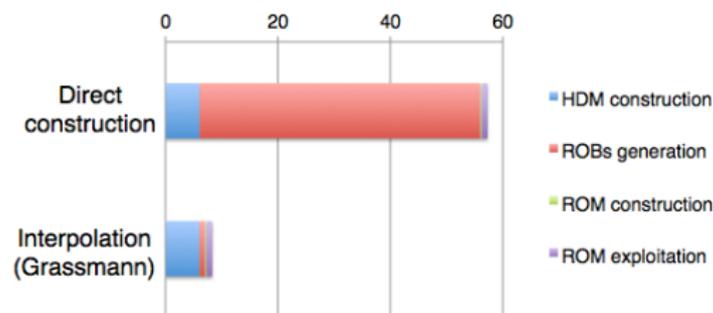
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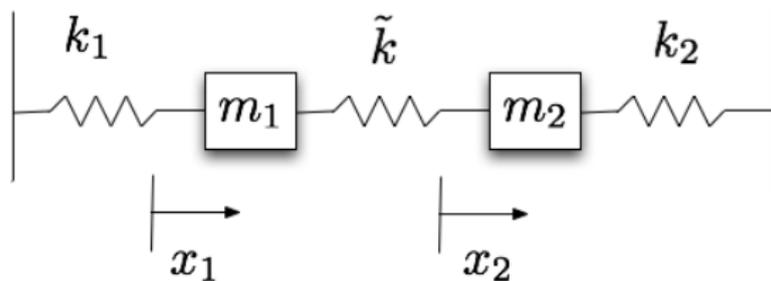
$$(\mathbf{A}(\boldsymbol{\mu}^*), \mathbf{B}(\boldsymbol{\mu}^*), \mathbf{C}(\boldsymbol{\mu}^*), \mathbf{D}(\boldsymbol{\mu}^*))$$

⇒ alternative approach is to interpolate the reduced-order operators

$$(\mathbf{A}_r(\boldsymbol{\mu}^{(l)}), \mathbf{B}_r(\boldsymbol{\mu}^{(l)}), \mathbf{C}_r(\boldsymbol{\mu}^{(l)}), \mathbf{D}_r(\boldsymbol{\mu}^{(l)}))$$

- Simple example: Mass-spring system with two degrees of freedom

$$\mathbf{M} \frac{d^2 \mathbf{w}}{dt^2}(t) + \mathbf{K}(\mu) \mathbf{w}(t) = \mathbf{B} \mathbf{u}(t), \quad \mu = k_1 - 0.1$$



$$\mathbf{w}(t) = \begin{bmatrix} x_1(t) \\ x_2(t) \end{bmatrix}$$

- Projection-based model order reduction by modal truncation: $\mathbf{V}(\boldsymbol{\mu})$ is the matrix of the two eigenmodes of the structural system

$$\mathbf{K}(\boldsymbol{\mu})\mathbf{v}_j(\boldsymbol{\mu}) = \lambda_j(\boldsymbol{\mu})\mathbf{M}\mathbf{v}_j(\boldsymbol{\mu})$$

└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

└ Application: Structural Analysis of a Simple Mass-Spring System

- Projection-based model order reduction by modal truncation: $\mathbf{V}(\boldsymbol{\mu})$ is the matrix of the two eigenmodes of the structural system

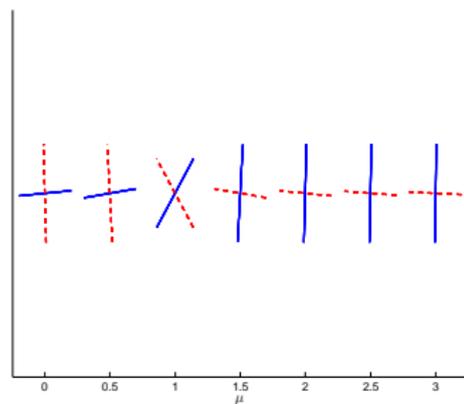
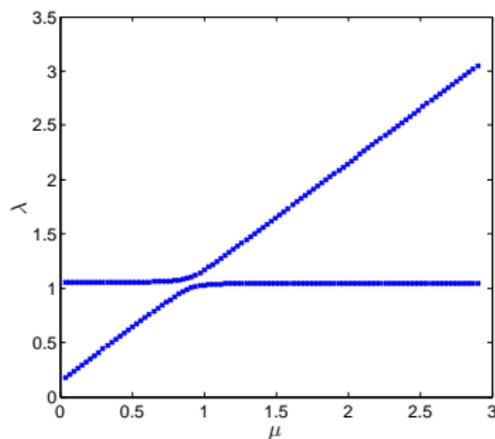
$$\mathbf{K}(\boldsymbol{\mu})\mathbf{v}_j(\boldsymbol{\mu}) = \lambda_j(\boldsymbol{\mu})\mathbf{M}\mathbf{v}_j(\boldsymbol{\mu})$$

- Matrix of eigenvalues: $\mathbf{K}_r(\boldsymbol{\mu}) = \mathbf{V}(\boldsymbol{\mu})^T \mathbf{K}(\boldsymbol{\mu}) \mathbf{V}(\boldsymbol{\mu}) = \boldsymbol{\Lambda}(\boldsymbol{\mu})$

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- Variations of the eigenvalues and eigenmodes with the parameter μ (first eigenmode is shown in blue color, second is shown in red color)



- Note that $\mathbf{\Lambda}(\boldsymbol{\mu})$ belongs to the manifold of (diagonal) symmetric positive definite matrices

└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

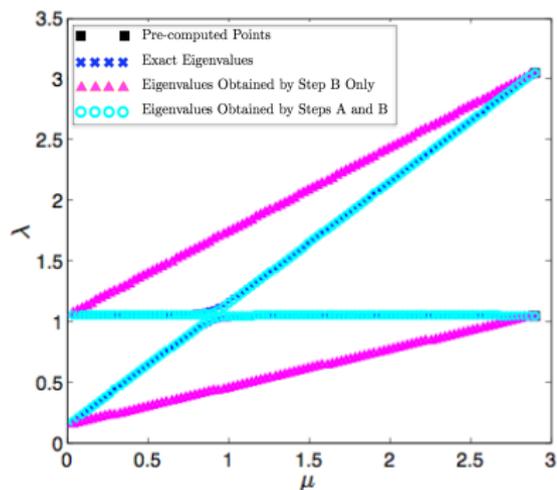
└ Interpolation on a Matrix Manifold

- Note that $\mathbf{\Lambda}(\boldsymbol{\mu})$ belongs to the manifold of (diagonal) symmetric positive definite matrices
- Perform interpolation of $\mathbf{\Lambda}(\boldsymbol{\mu})$ on this manifold using $(\mathbf{\Lambda}(0), \mathbf{\Lambda}(2.9))$

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- Interpolation on a Matrix Manifold

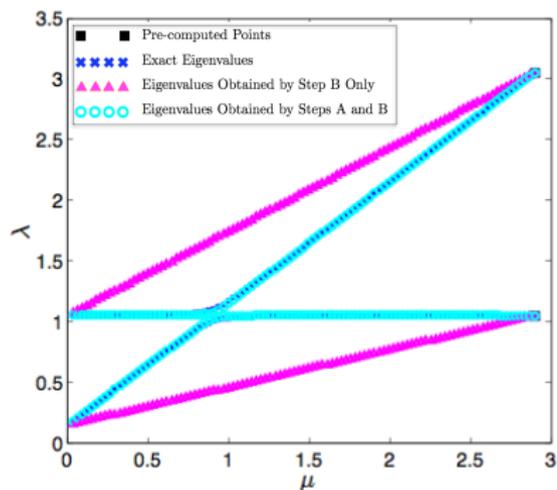
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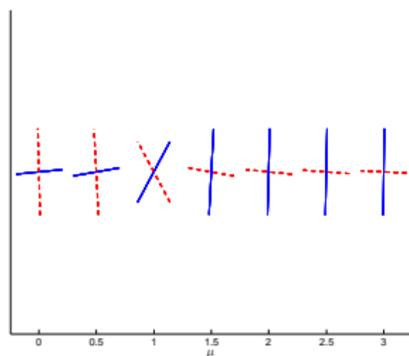
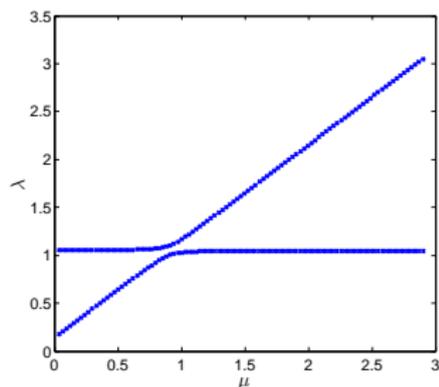
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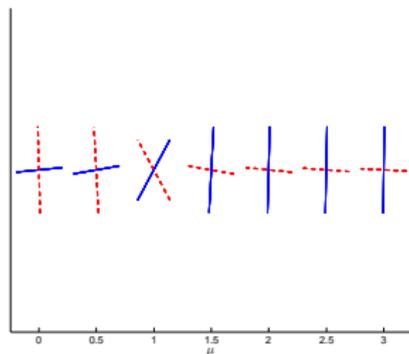
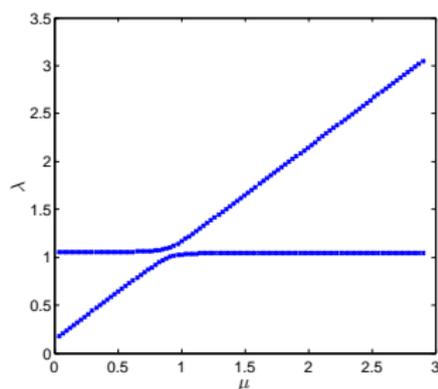


- Observe that the result is wrong, even for such a simple system

- The issue is the lack of consistency between the coordinates of the reduced-order matrices, triggered in this case by **mode veering**



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- Mode crossing would trigger a similar issue (the eigenfrequencies are ordered increasingly in $\mathbf{\Lambda}(\mu)$)

Two-step solution

- step A: Pre-process the reduced-order matrices
 - enforce consistency by solving the following N_s **orthogonal Procrustes problems**

$$\min_{\mathbf{Q}_l \mathbf{Q}_l^T \mathbf{Q}_l = \mathbf{I}_k} \left\| \mathbf{V}(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l - \mathbf{V}(\boldsymbol{\mu}^{(b)}) \right\|_F, \quad \forall l = 1, \dots, N_s$$

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- compute analytical solutions of above problems as follows
 - 1 compute $\mathbf{P}_{l,l_0} = \mathbf{V}(\boldsymbol{\mu}^{(l)})^T \mathbf{V}(\boldsymbol{\mu}^{(l_0)})$
 - 2 compute the SVD $\mathbf{P}_{l,l_0} = \mathbf{U}_{l,l_0} \boldsymbol{\Sigma}_{l,l_0} \mathbf{Z}_{l,l_0}^T$
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- the associated computational cost scales with k

⇒ step A can be performed either online or offline

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- step A: Pre-process the reduced-order matrices
 - enforce consistency by solving the following N_s **orthogonal Procrustes problems**

$$\min_{\mathbf{Q}_l \mathbf{Q}_l^T = \mathbf{I}_k} \left\| \mathbf{V}(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l - \mathbf{V}(\boldsymbol{\mu}^{(l_0)}) \right\|_F, \quad \forall l = 1, \dots, N_s$$

- compute analytical solutions of above problems as follows
 - 1 compute $\mathbf{P}_{l,l_0} = \mathbf{V}(\boldsymbol{\mu}^{(l)})^T \mathbf{V}(\boldsymbol{\mu}^{(l_0)})$
 - 2 compute the SVD $\mathbf{P}_{l,l_0} = \mathbf{U}_{l,l_0} \boldsymbol{\Sigma}_{l,l_0} \mathbf{Z}_{l,l_0}^T$
 - 3 compute $\mathbf{Q}_l = \mathbf{U}_{l,l_0} \mathbf{Z}_{l,l_0}^T$
- the associated computational cost scales with k

⇒ step A can be performed either online or offline

Two-step solution (continue)

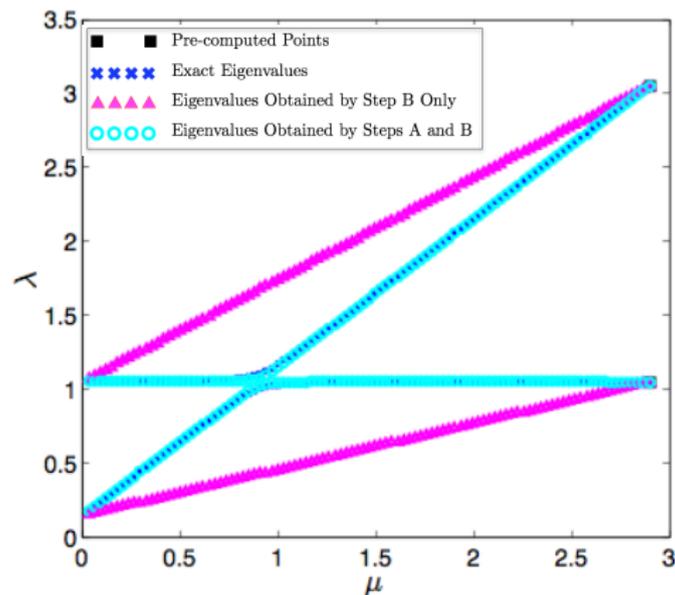
- step B: Note that (assuming a Galerkin PROM and orthogonal local ROBs)

$$\begin{aligned} (\mathbf{V}(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l)^T \mathbf{A}(\boldsymbol{\mu}^{(l)}) (\mathbf{V}(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l) &= \mathbf{Q}_l^T \mathbf{V}(\boldsymbol{\mu}^{(l)})^T \mathbf{A}(\boldsymbol{\mu}^{(l)}) \mathbf{V}(\boldsymbol{\mu}^{(l)}) = \mathbf{Q}_l^T \mathbf{A}_r(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l \\ (\mathbf{V}(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l)^T \mathbf{B}(\boldsymbol{\mu}^{(l)}) &= \mathbf{Q}_l^T \mathbf{V}(\boldsymbol{\mu}^{(l)})^T \mathbf{B}(\boldsymbol{\mu}^{(l)}) = \mathbf{Q}_l^T \mathbf{B}_r(\boldsymbol{\mu}^{(l)}) \\ \mathbf{C}(\boldsymbol{\mu}^{(l)}) (\mathbf{V}(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l) &= (\mathbf{C}(\boldsymbol{\mu}^{(l)}) \mathbf{V}(\boldsymbol{\mu}^{(l)})) \mathbf{Q}_l = \mathbf{C}_r \mathbf{Q}_l \end{aligned}$$

and therefore

- first, transform *directly* each PROM $(\mathbf{A}_r(\boldsymbol{\mu}^{(l)}), \mathbf{B}_r(\boldsymbol{\mu}^{(l)}), \mathbf{C}_r(\boldsymbol{\mu}^{(l)}), \mathbf{D}_r(\boldsymbol{\mu}^{(l)}))$ to $(\mathbf{Q}_l^T \mathbf{A}_r(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l, \mathbf{Q}_l^T \mathbf{B}_r(\boldsymbol{\mu}^{(l)}), \mathbf{C}_r(\boldsymbol{\mu}^{(l)}) \mathbf{Q}_l, \mathbf{D}_r(\boldsymbol{\mu}^{(l)}))$
- then, identify for each element of the transformed tuple an appropriate matrix manifold and perform the interpolation on this matrix manifold

- Result is shown in cyan color

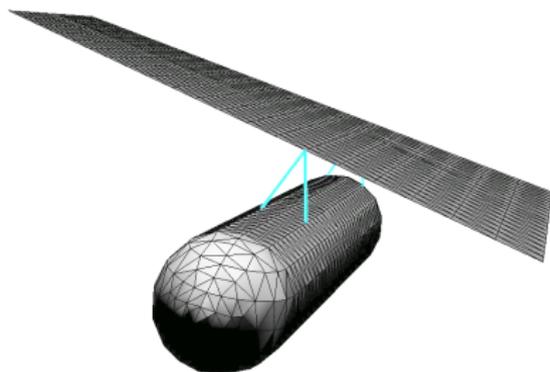


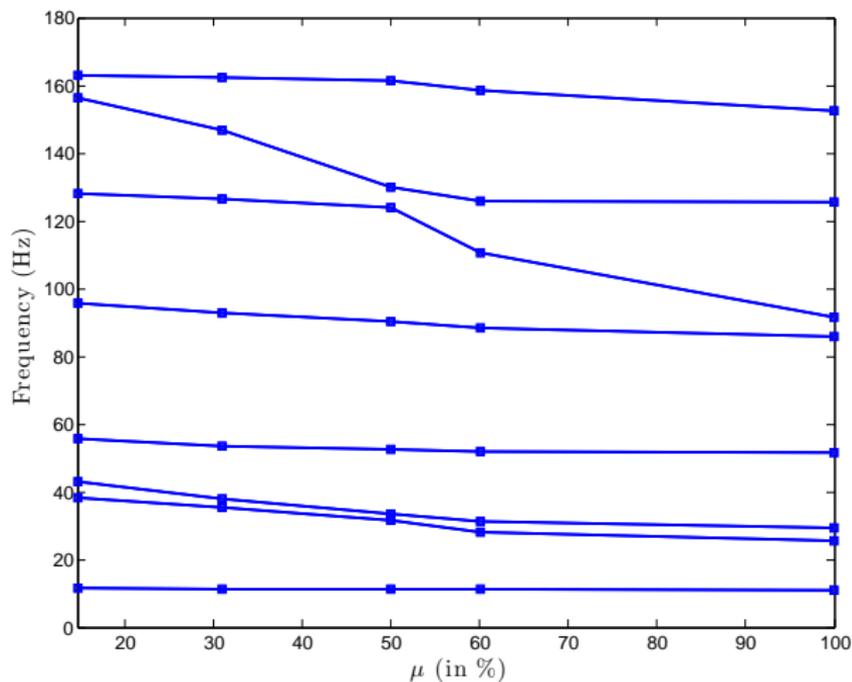
- Observe that the result is very accurate

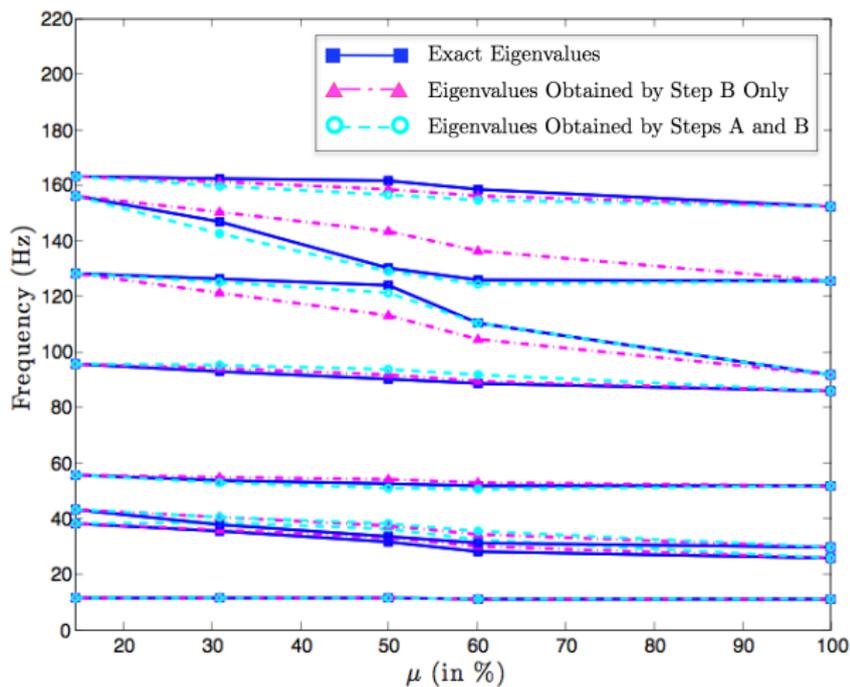
└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

└ Application: Structural Analysis of a Wing-Tank Configuration

- More challenging example: Wing with tank and sloshing effects
- The hydro-elastic effects affect the eigenfrequencies and eigenmodes of the structure
- The parameter μ defines the fuel fill level in the tank $0 \leq \mu \leq 100\%$







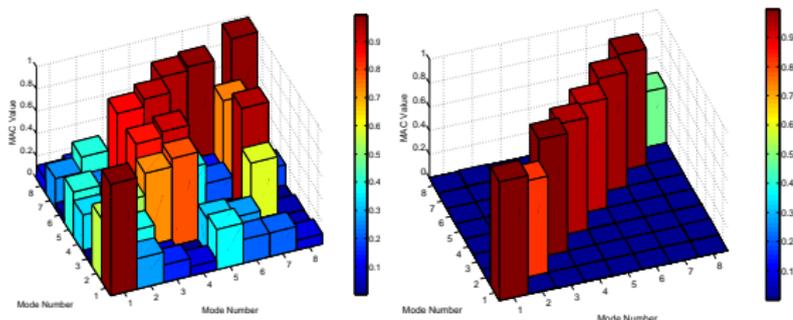
- Modal Assurance Criterion (MAC) between two modes ϕ and ψ

$$\text{MAC}(\phi, \psi) = \frac{|\phi^T \psi|^2}{(\phi^T \phi)(\psi^T \psi)}$$

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- What is the MAC between the vectors in the ROBs $\mathbf{V}(\mu^{(l)})$ and $\mathbf{V}(\mu^{(l_0)})$ before and after Step A?



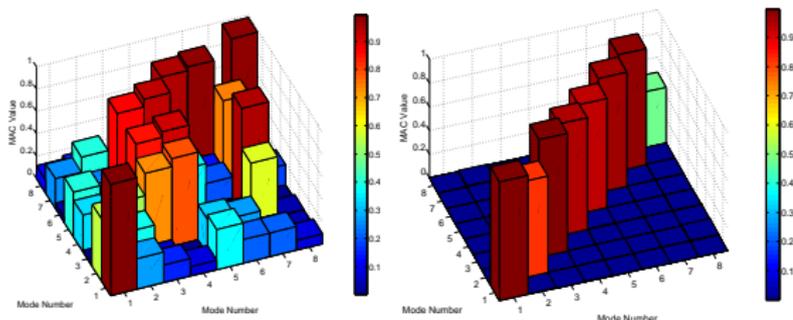
└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

└ Link with Modal Assurance Criterion

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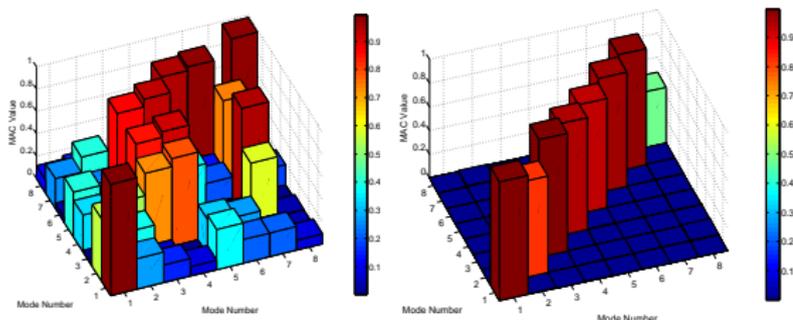


- When ϕ and ψ are normalized, $\text{MAC}(\phi, \psi) = |\phi^T \psi|^2$

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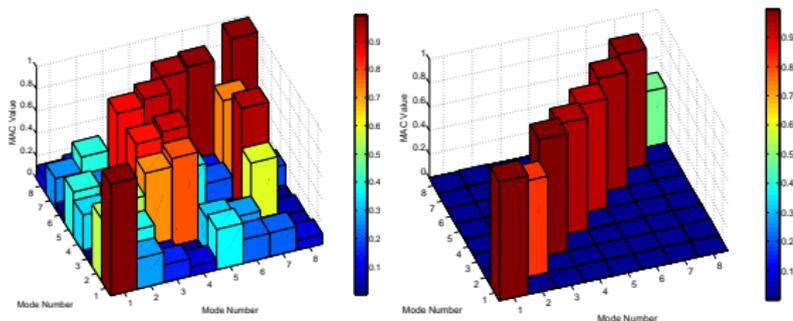


- When ϕ and ψ are normalized, $\text{MAC}(\phi, \psi) = |\phi^T \psi|^2$
- \mathbf{P}_{l,l_0} is the matrix of square roots of the MACs between the modes contained in $\mathbf{V}(\mu^{(l)})$ and those contained in $\mathbf{V}(\mu^{(l_0)})$

- Modal Assurance Criterion (MAC) between two modes ϕ and ψ

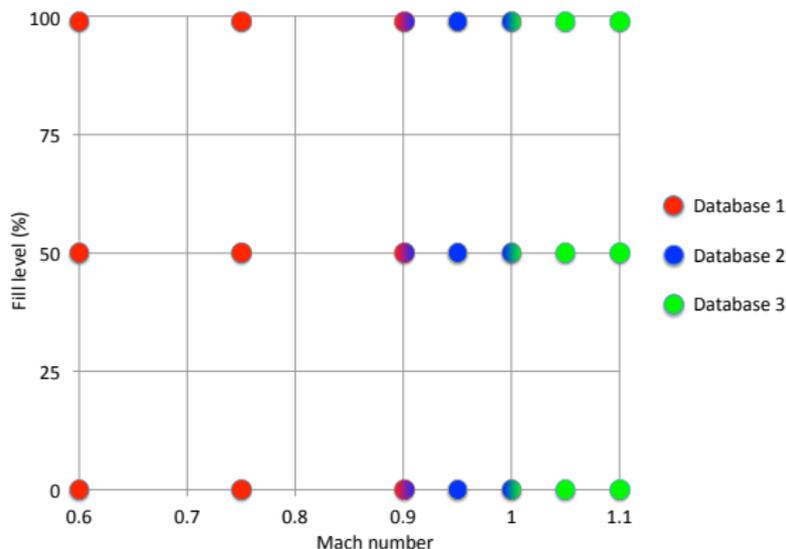
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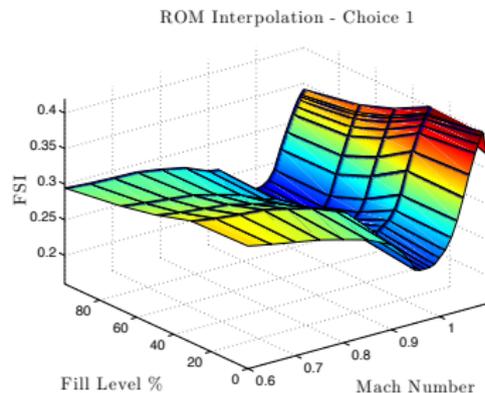
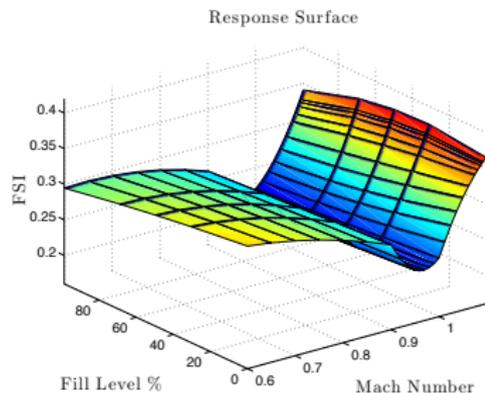
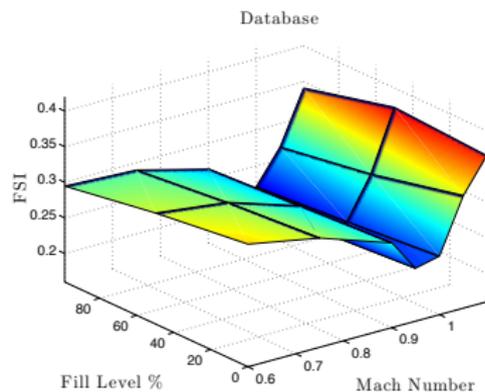
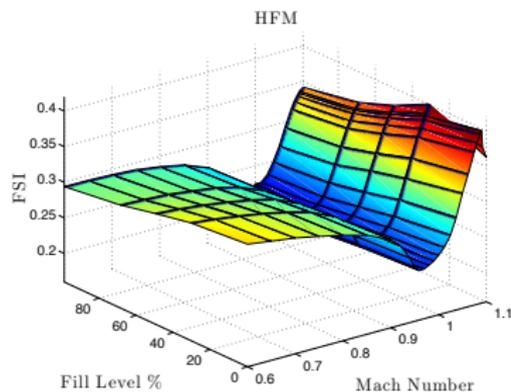
- When ϕ and ψ are normalized, $\text{MAC}(\phi, \psi) = |\phi^T \psi|^2$
- \mathbf{P}_{l,l_0} is the matrix of square roots of the MACs between the modes contained in $\mathbf{V}(\mu^{(l)})$ and those contained in $\mathbf{V}(\mu^{(l_0)})$
- This is the Modal Assurance Criterion Square Root (MACSR)

- Aeroelastic study of a wing-tank system
- 2 parameters, namely, the fuel fill level and the free-stream Mach number M_∞
- Database approach



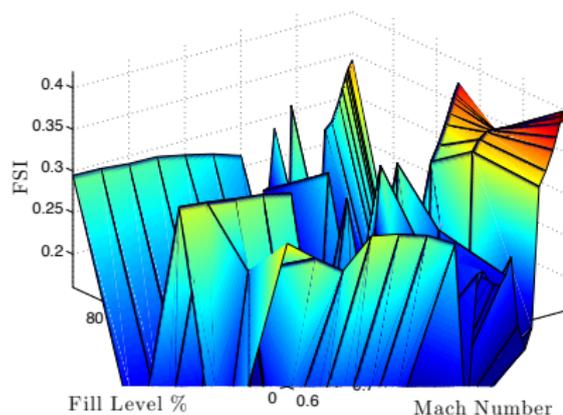
└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

└ Application: Aeroelastic Analysis of a Wing-Tank Configuration

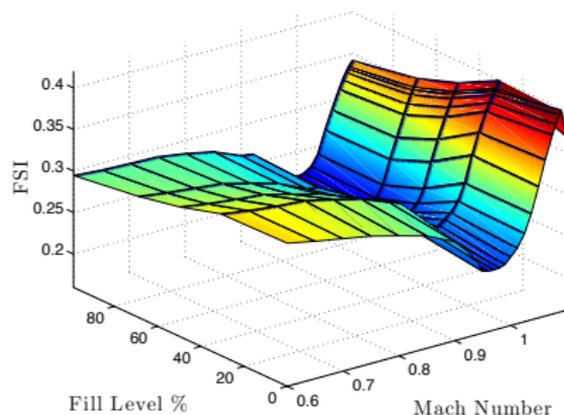


- Effect of Step A

ROM Interpolation - Choice 1



ROM Interpolation - Choice 1

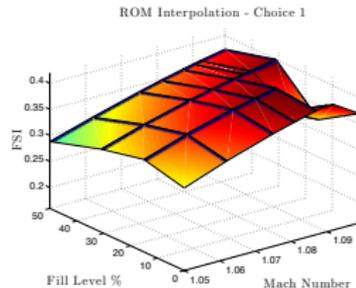
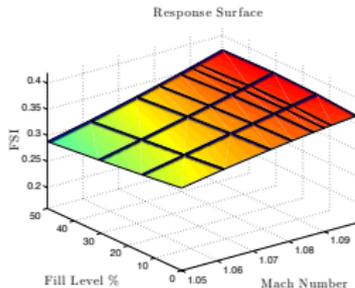
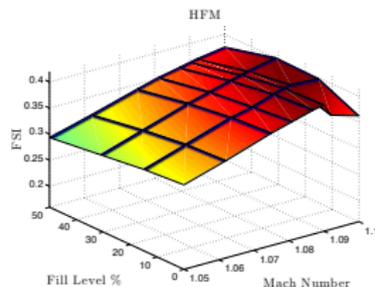


- Skipping Step A leads to inaccurate interpolation results (left figure)
- Step A ensures a consistent interpolation (right figure)

└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

└ Application: Aeroelastic Analysis of a Wing-Tank Configuration

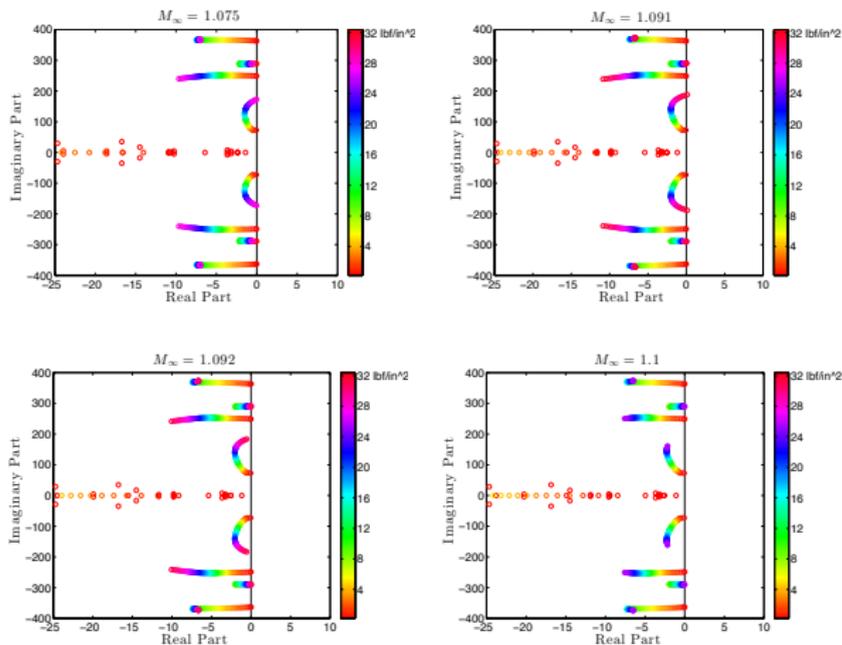
- The consistent interpolation on a matrix manifold is able to detect aeroelastic bifurcations



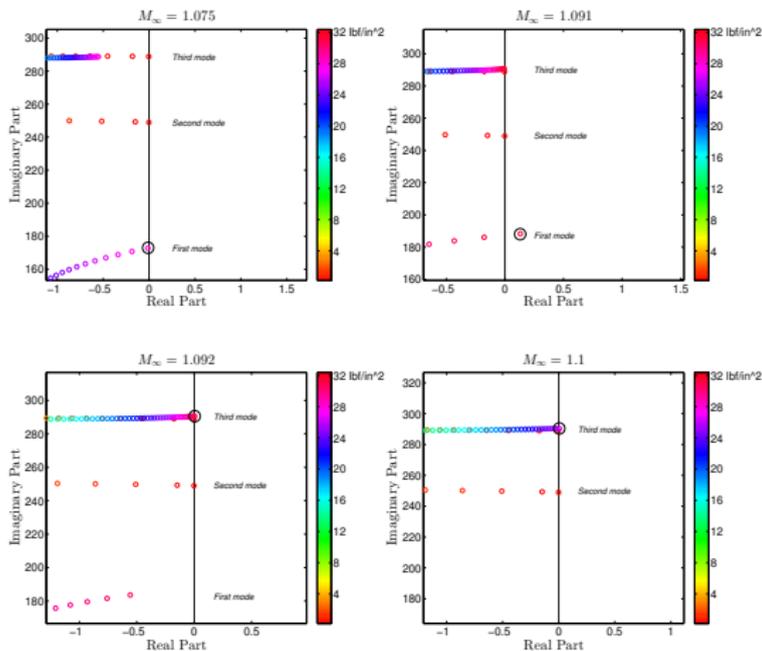
└ Interpolation of PROMs on Embedded Manifolds (Amsallem and Farhat, 2011)

└ Application: Aeroelastic Analysis of a Wing-Tank Configuration

- The consistent interpolation on a matrix manifold is able to detect aeroelastic bifurcations (0% fuel fill level)



- The consistent interpolation on a matrix manifold is able to detect aeroelastic bifurcations (0% fuel fill level)

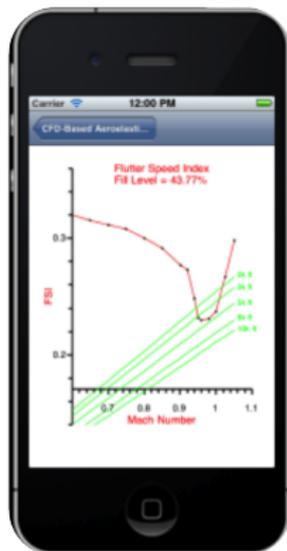


- CPU performance

Approach	Offline phase CPU time (# procs)	Online phase CPU time (# procs)
HDM	- (-)	9,152,000 s \approx 106 days (32)
Response Surface	28,000 s \approx 7 h (32)	2 s (1)
PROM Interpolation	28,000 s \approx 7 h (32)	30 s (1)

- Online speedup = 305,000x
- Offline+Online speedup = 327x

- Mobile computing using a database of PROMs



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- D. Amsallem, J. Cortial, C. Farhat. Towards real-time CFD-based aeroelastic computations using a database of reduced-order models. *AIAA Journal* 2010; 48(9):2029-2037.
- D. Amsallem, C. Farhat. An online method for interpolating linear parametric reduced-order models. *SIAM Journal on Scientific Computing* 2011; 33(5): 2169-2198.
- D. Amsallem, Interpolation on manifolds of CFD-based fluid and finite element-based structural reduced-order models for on-line aeroelastic predictions. Ph.D. Thesis, Stanford University, 2010.